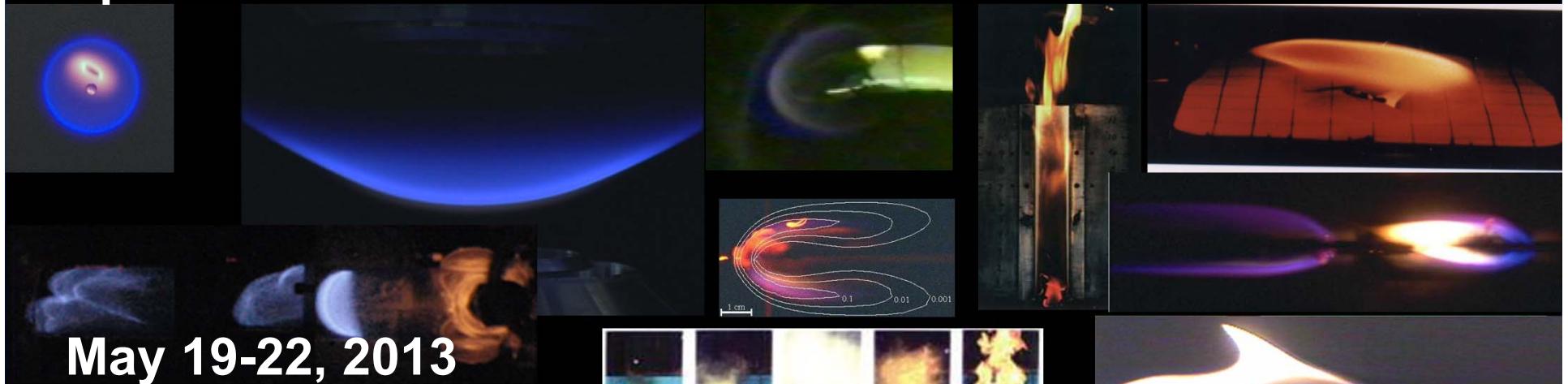
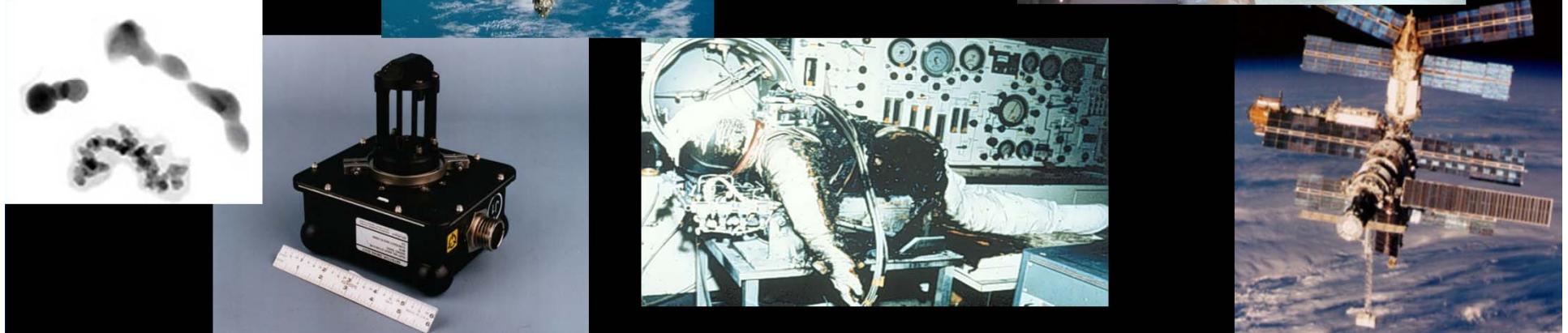
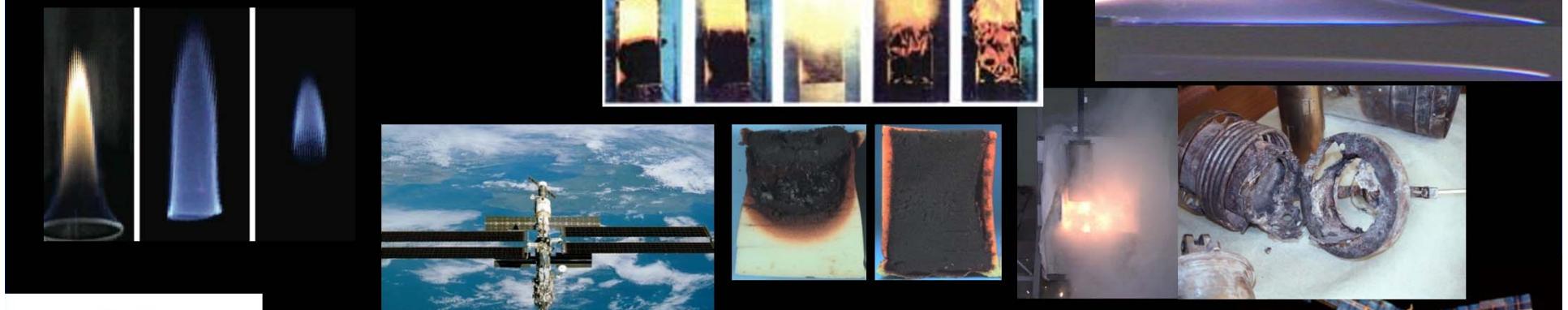


Evaluation of Low-Gravity Smoke Particulate for Spacecraft Fire Detection



May 19-22, 2013





Evaluation of Low-Gravity Smoke Particulate for Spacecraft Fire Detection

NASA Glenn Research Center

**David Urban, Gary Ruff, Paul Greenberg,
Marit Meyer**

University of Maryland

George Mulholland

National Center for Space Exploration Research

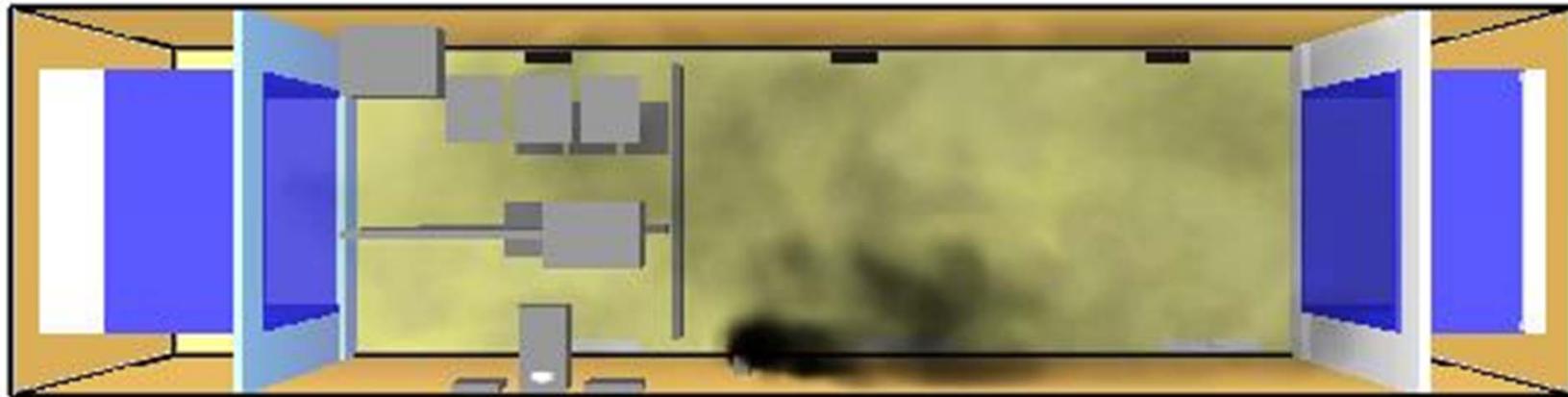
Zeng-guang Yuan, Victoria Bryg

National Institute of Standards and Technology

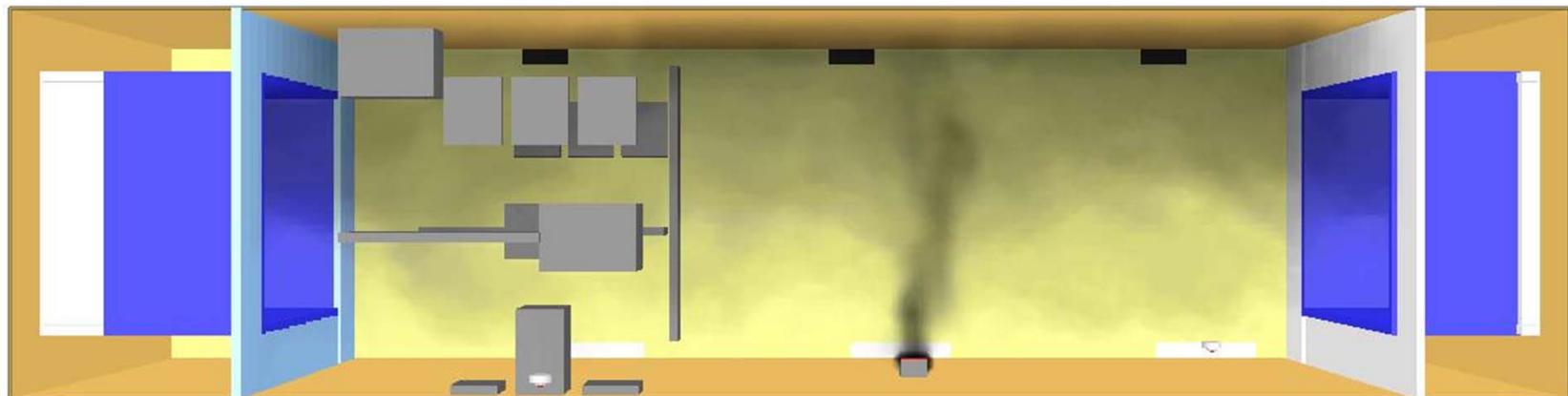
Thomas Cleary, Jiann Yang



Smoke Detection Background: Destiny Smoke Detection Simulation-25% Soot effect of gravity



Low-gravity



Normal-gravity



Background: Spacecraft Fire Detection

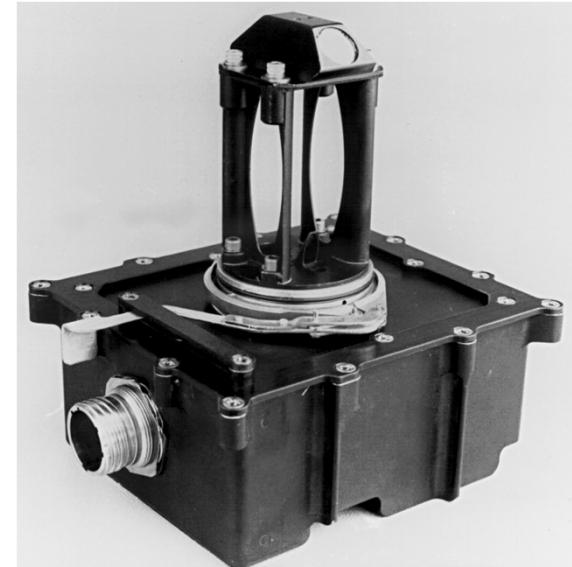


STS Detector: sensitive < 1 micron

- dual-chamber ionization with inertial separator which rejects particles larger than 1-2 microns.
- Developed in the late 70's when Ionization detectors were prevalent

ISS detector: sensitive > 0.5 micron

- 2-pass IR laser-diode forward-scattering detector (30 degrees) minimum reported sensitivity is $0.3 \mu\text{m}$.
- Developed in the 90's and took advantage of the availability of stable diode light sources.



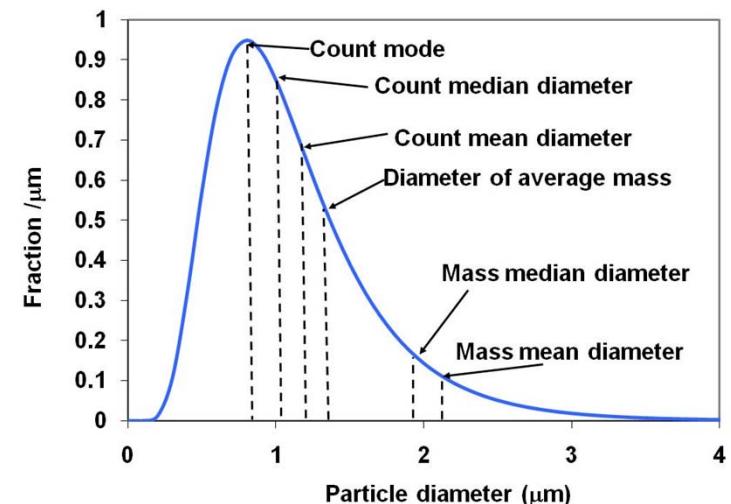
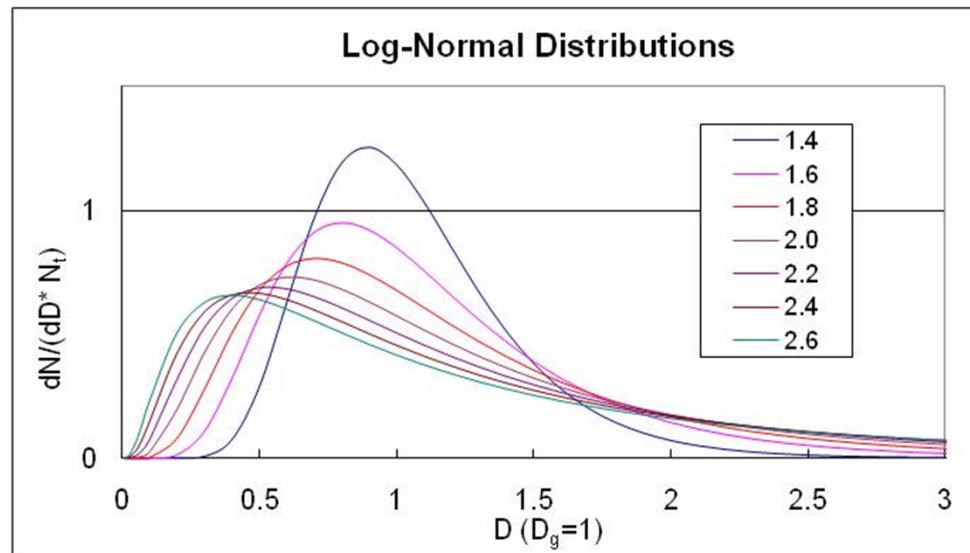


Log-Normal Distribution

$$f_N(D) = \frac{N_t}{(2\pi)^{1/2} D \ln \sigma_g} \exp\left(-\frac{(\ln D - \ln D_g)^2}{2 \ln^2 \sigma_g}\right)$$

Number is dominated by the smaller particles

Mass is dominated by the larger particles (tail)



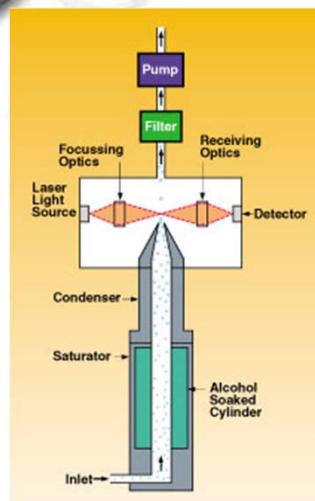
Log-normal distribution $\sigma_g = 1.6$, $D_g = 1$



SAME Experimental Diagnostic Measurements

All measure moments of the particle size distribution

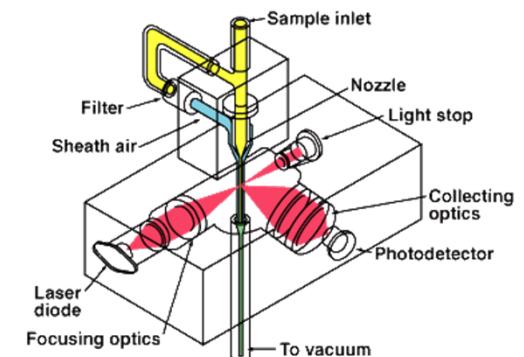
$$M_i = \int D^i f_N(D) dD$$



Zeroth Moment:
TSI PTrak™



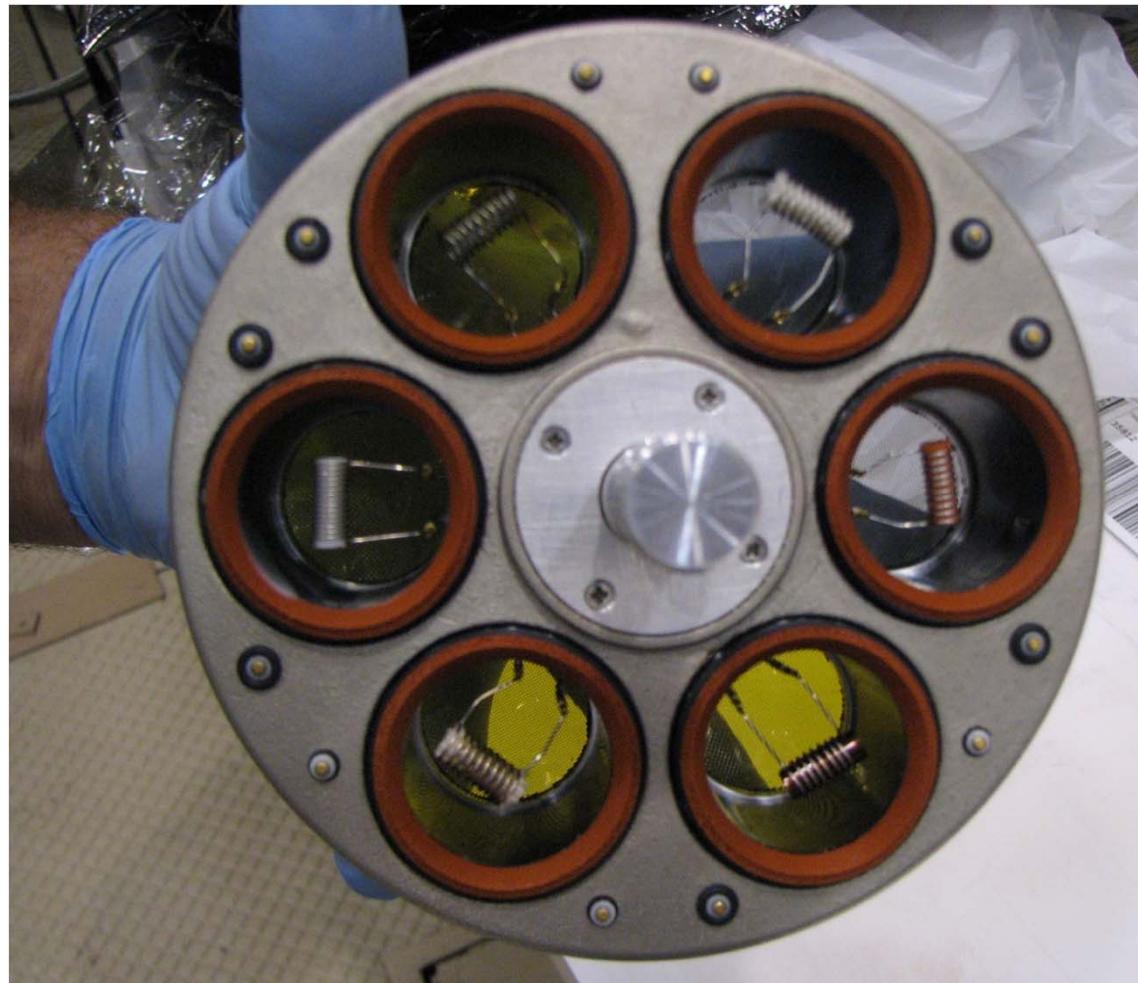
First Moment:
First Alert™
Smoke Detector



Third Moment:
TSI Dust Trak™



SAME Sample Carousel



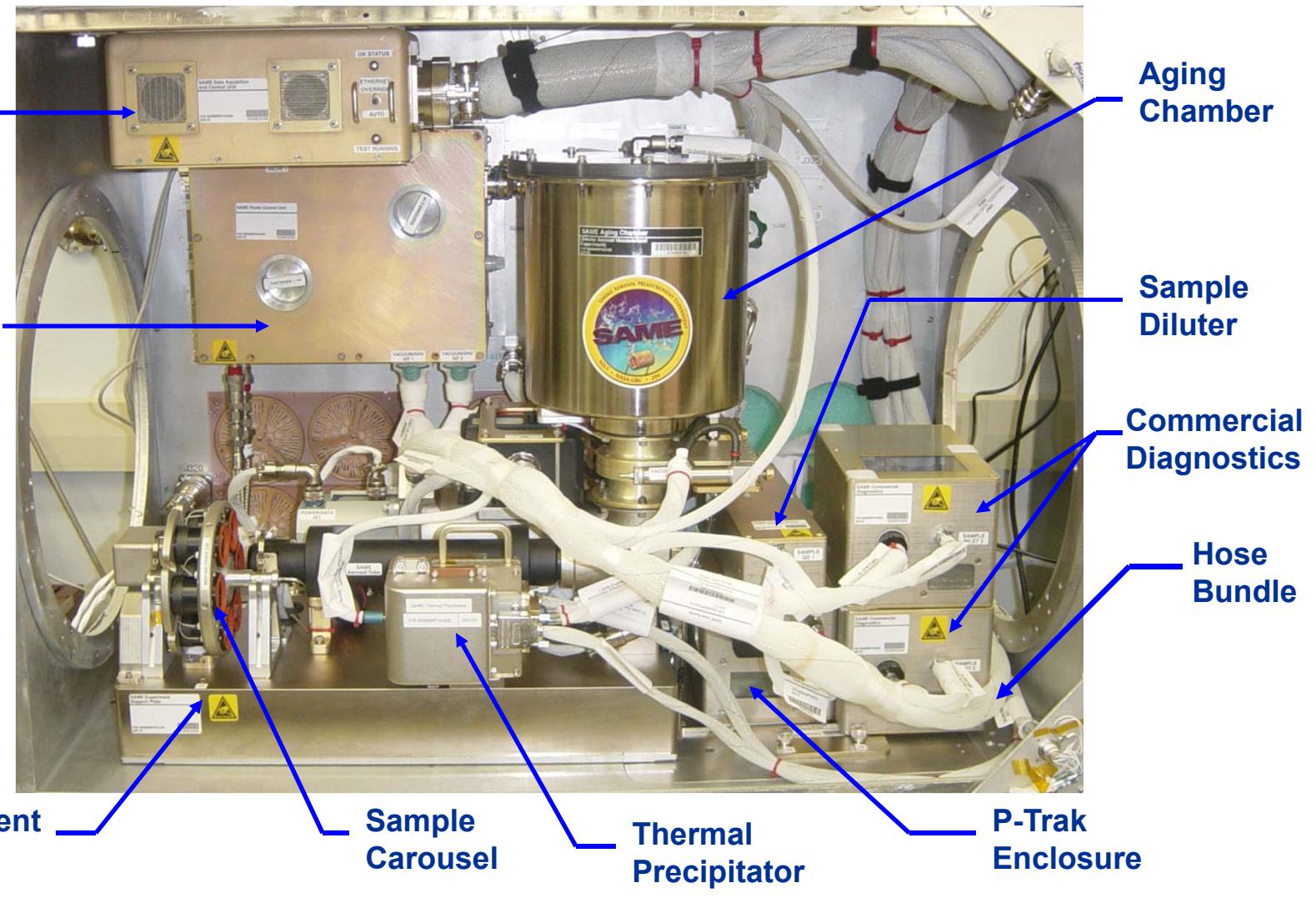
Sample Materials:

**Silicone
Teflon
Kapton
LampWick
Pyrell
DBP**

SAME in MSG (mockup)



Data
Acquisition and
Control Unit



Experiment
Support
Plate

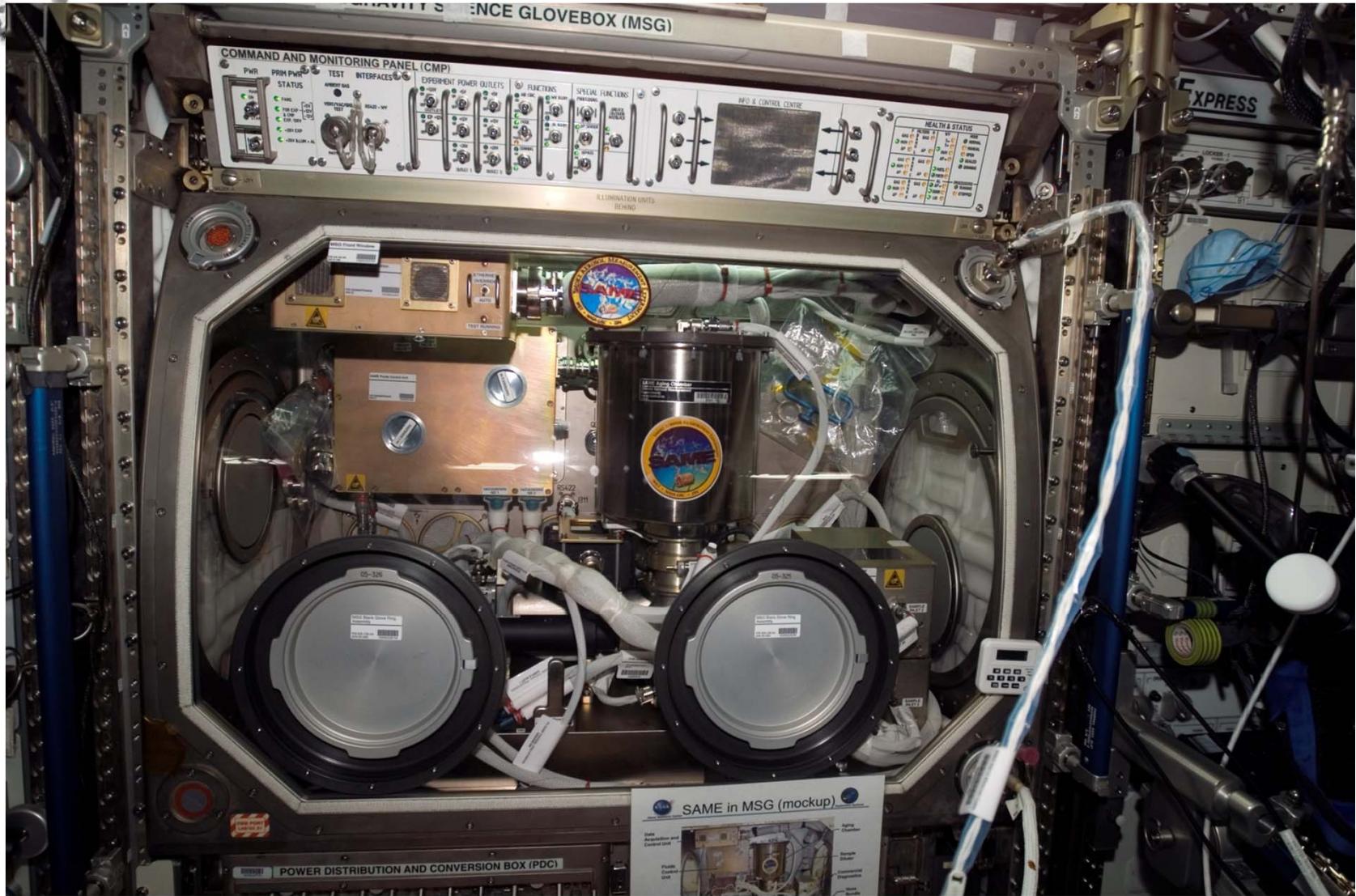
Sample
Carousel

Thermal
Precipitator

P-Trak
Enclosure

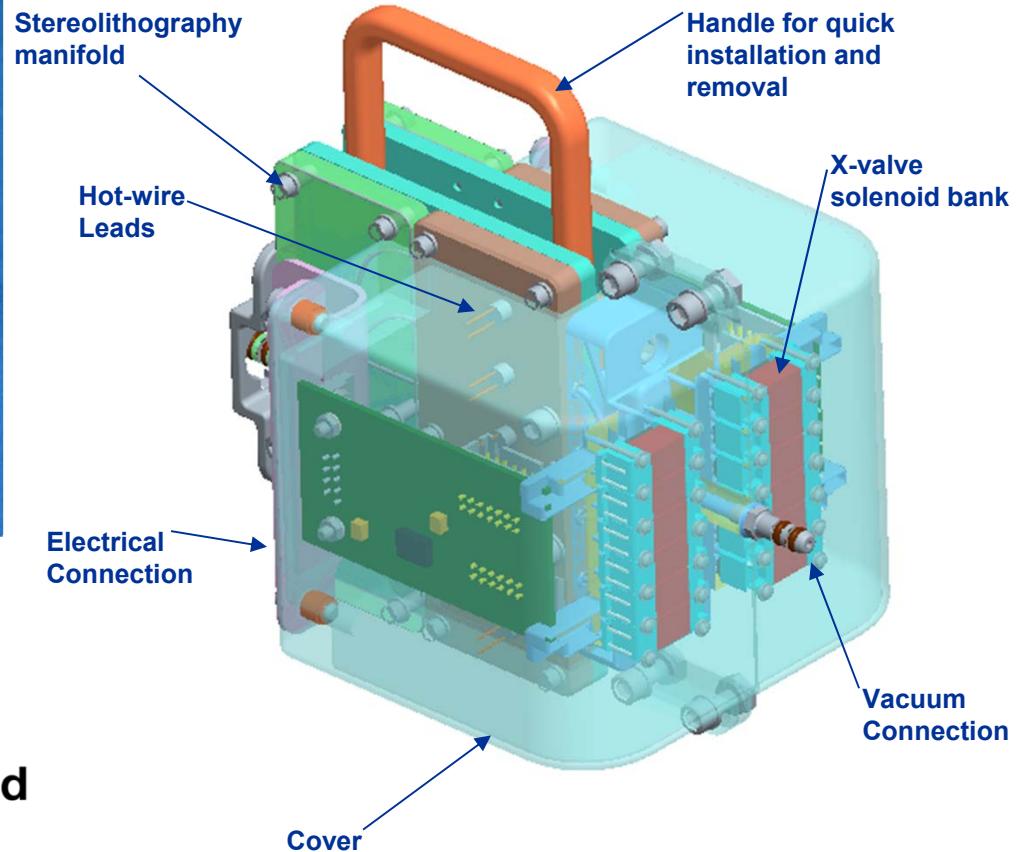
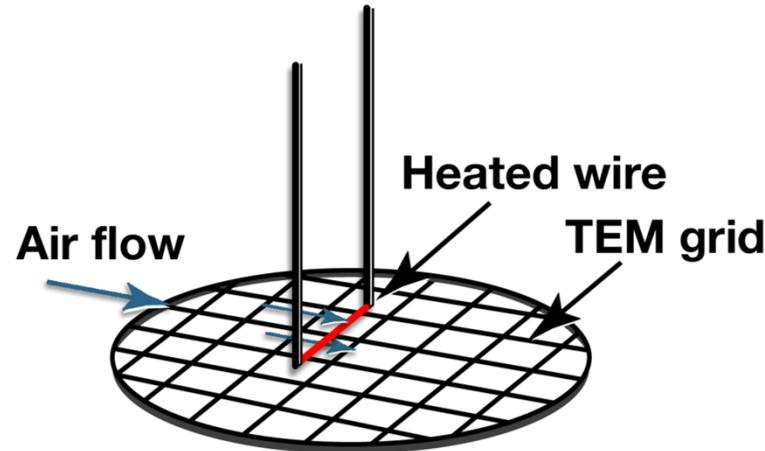


SAME Hardware on orbit



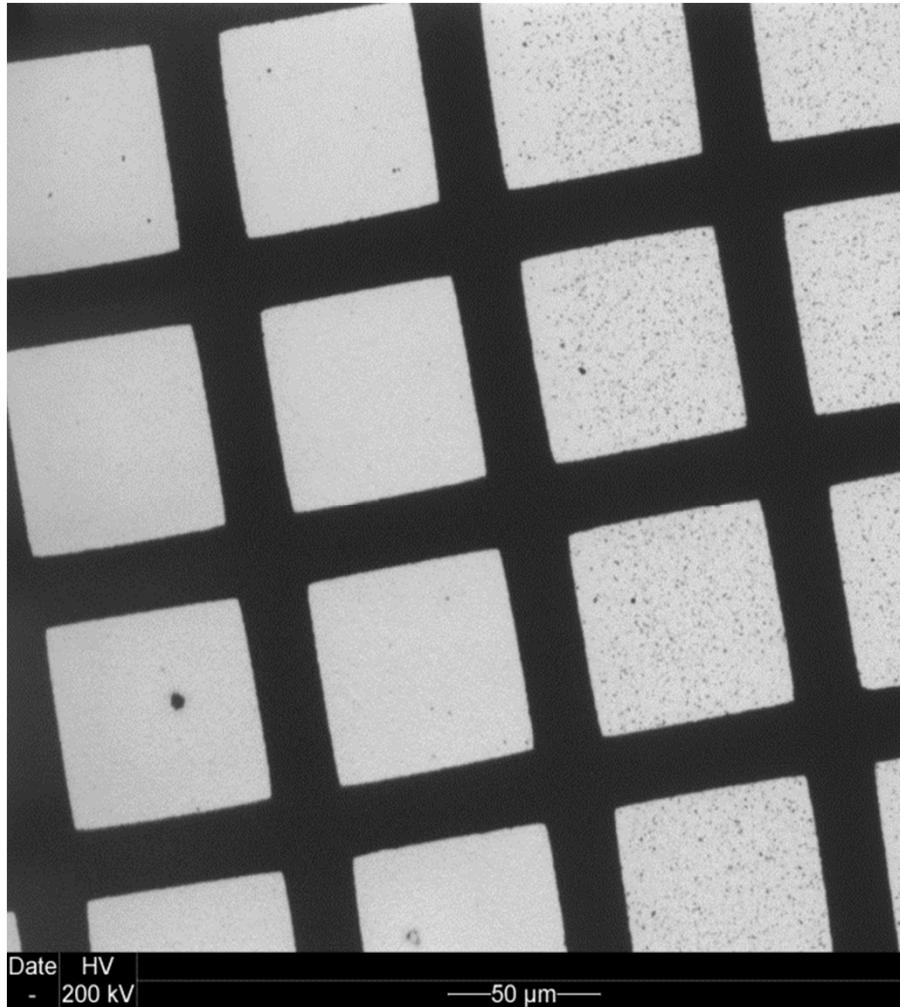
ISS015E26265

SAME Particle Capture





Thermal Precipitator



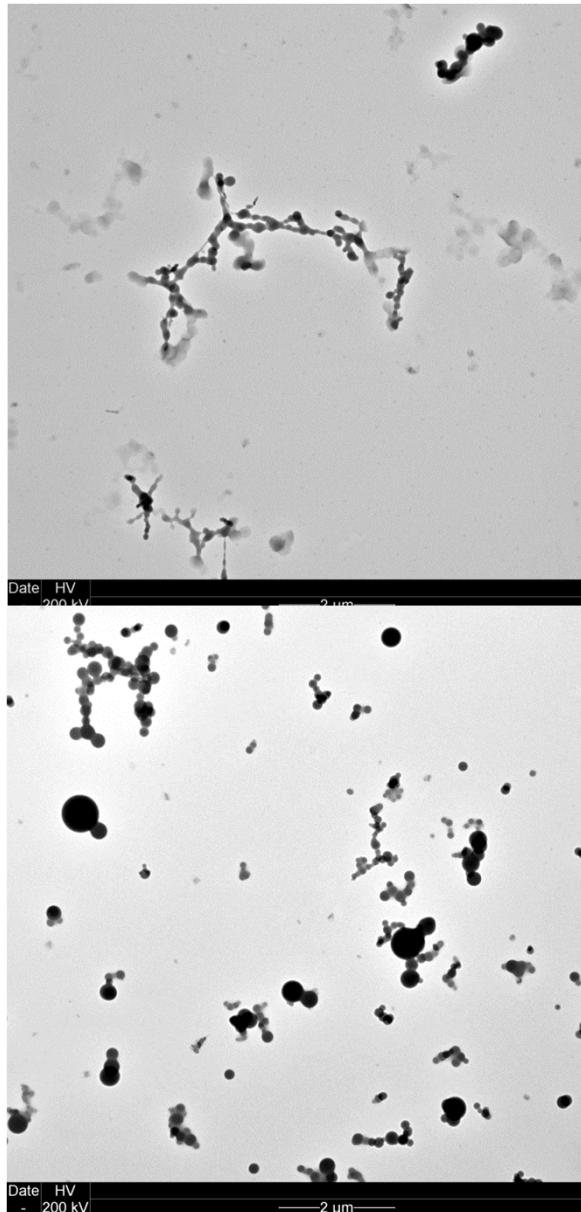
Overview image showing deposition boundary



TEM Results

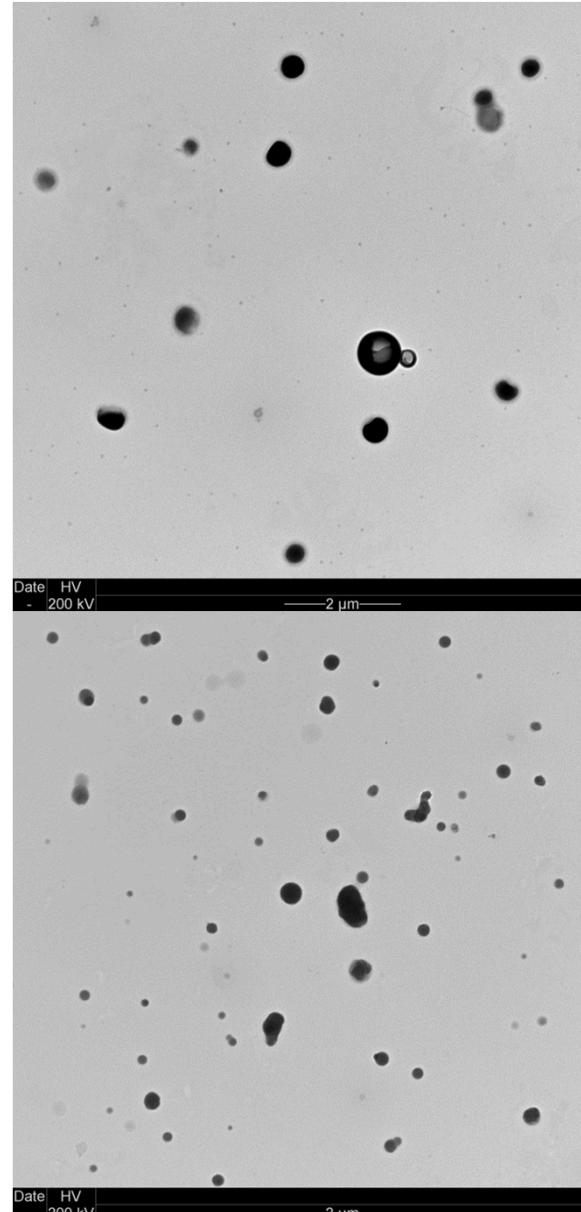


Teflon
(Run 56)



Pyrell
(Run 63)

Length scale
2 μm

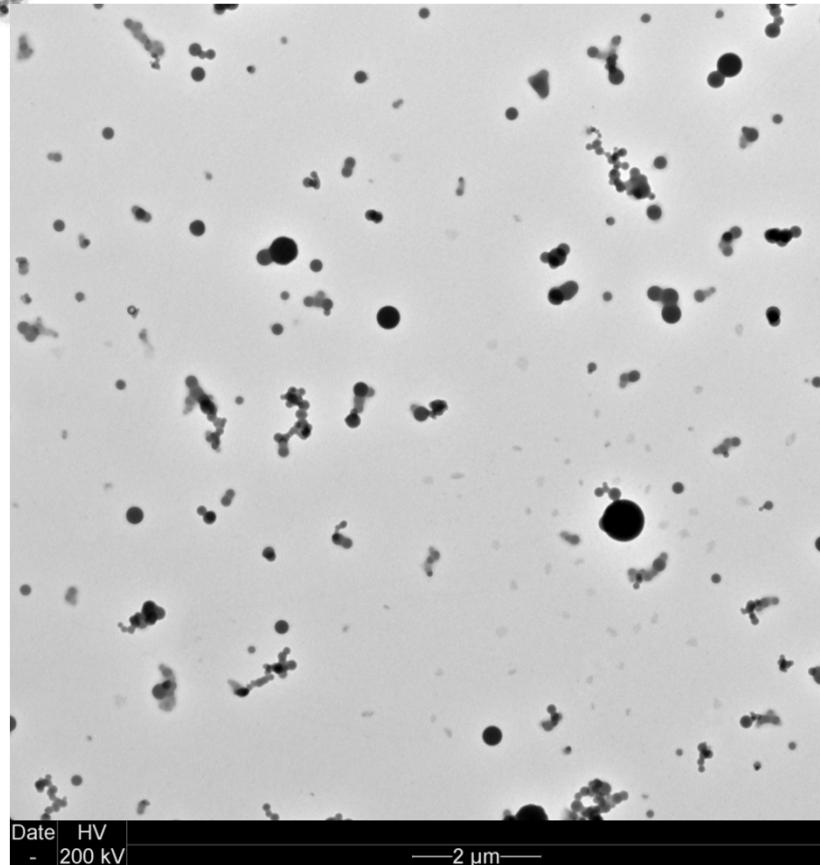


Lampwick
(Run 54)

Kapton
(Run
62)

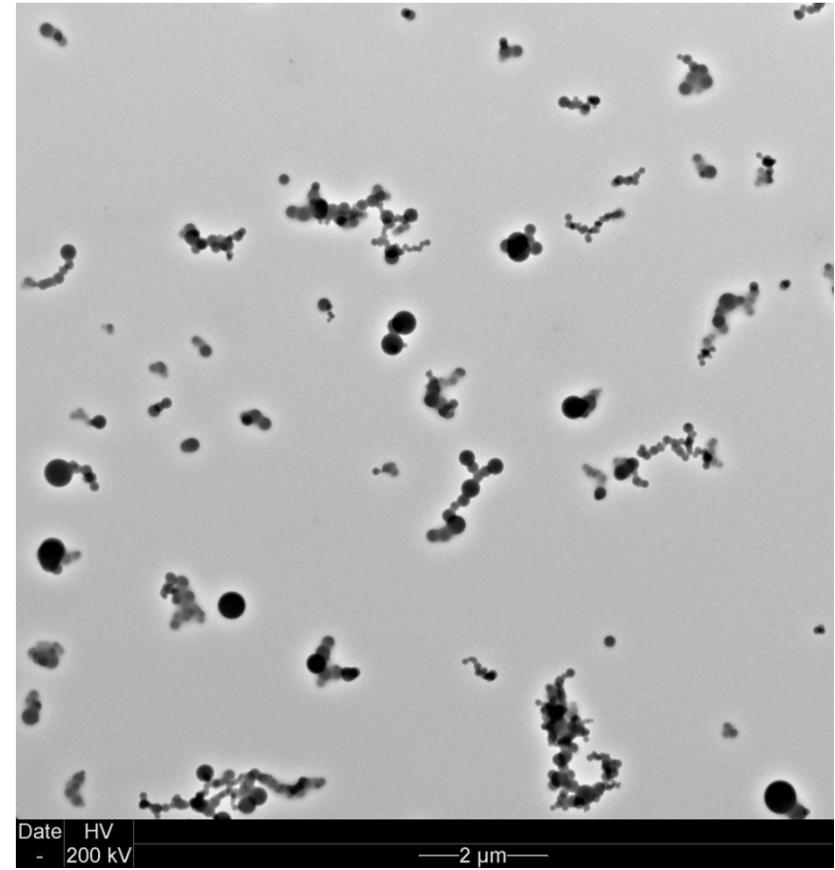


TEM Results - Pyrell Aging



Pre aging

2 microns



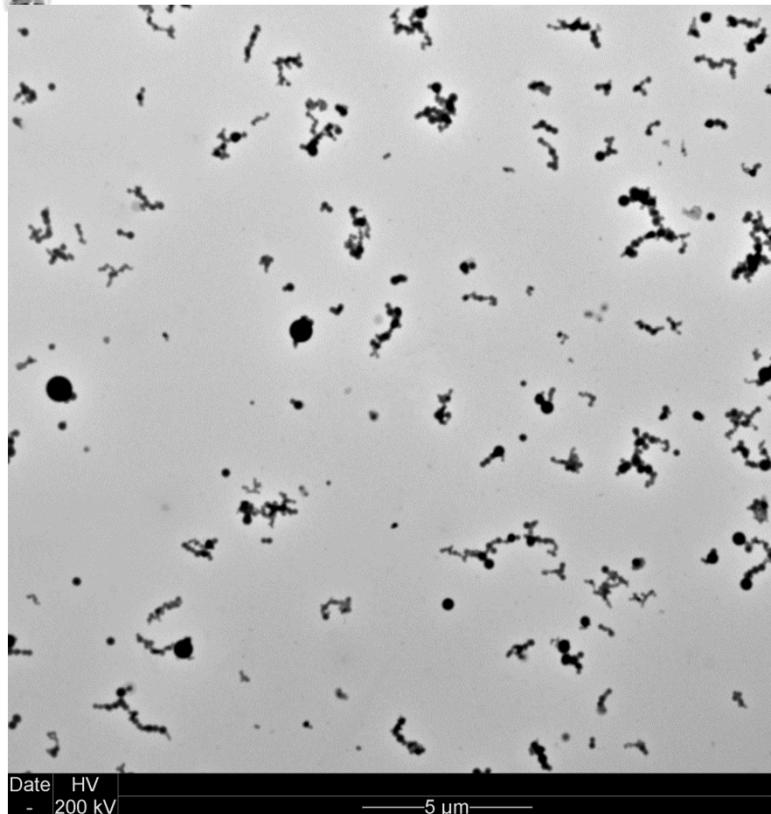
Post aging

2 microns

High Temperature Pyrell: 480 second aging run (Run 84)



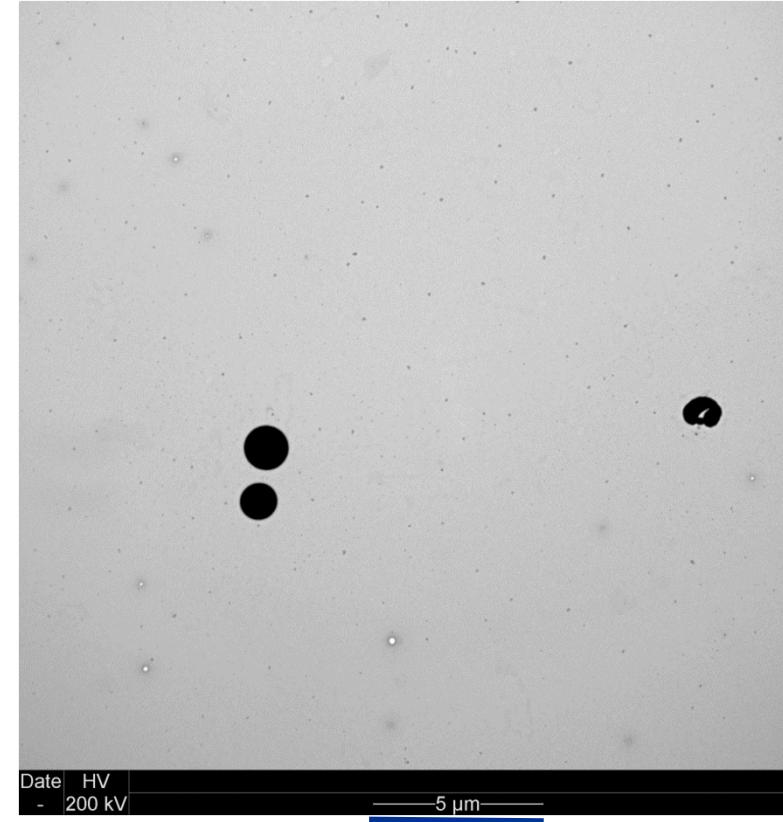
TEM Results – Pyrell- effect of flow



5 microns

8 cm/s air flow

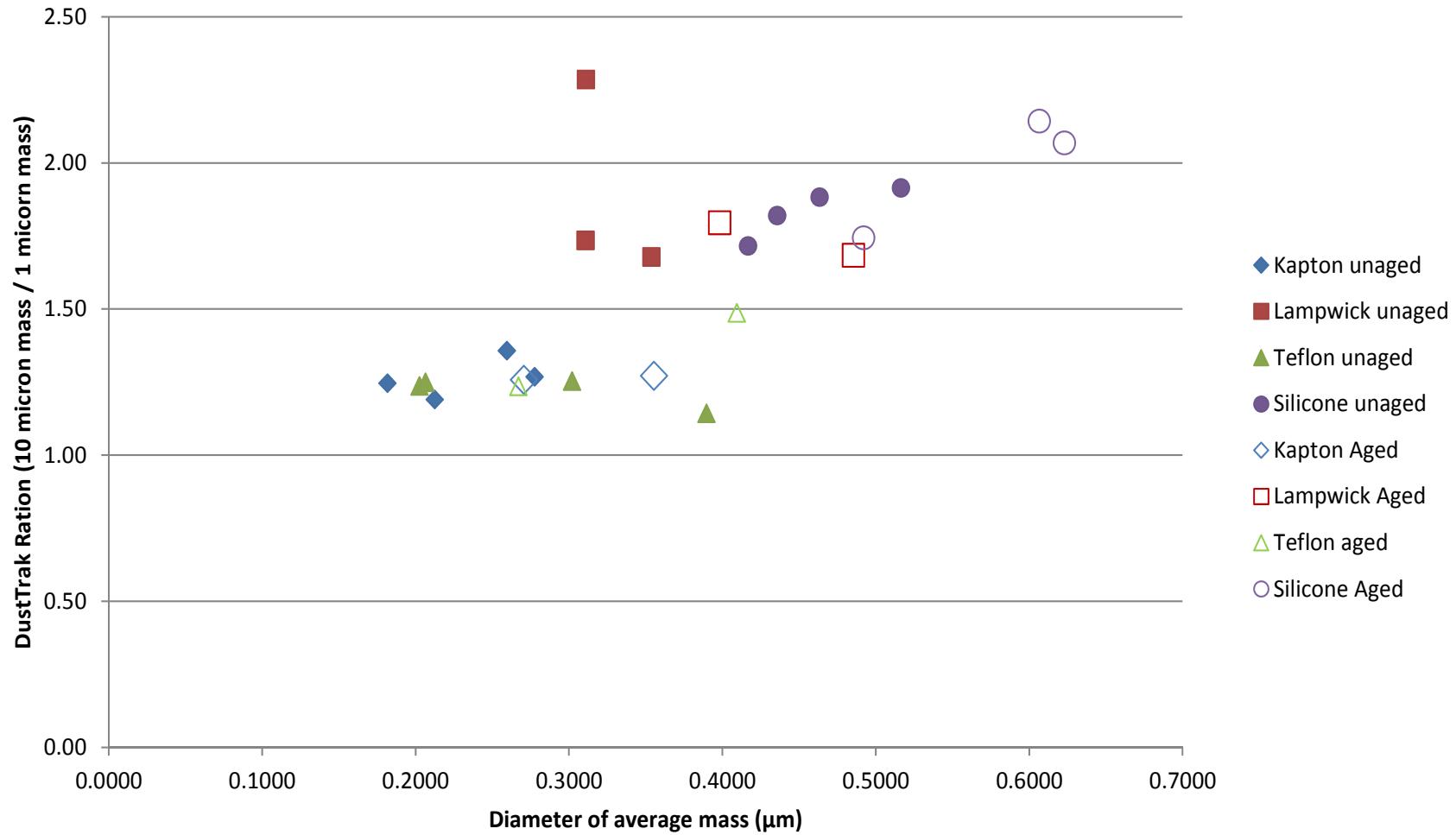
Pyrell with and without flow



5 microns

No air flow

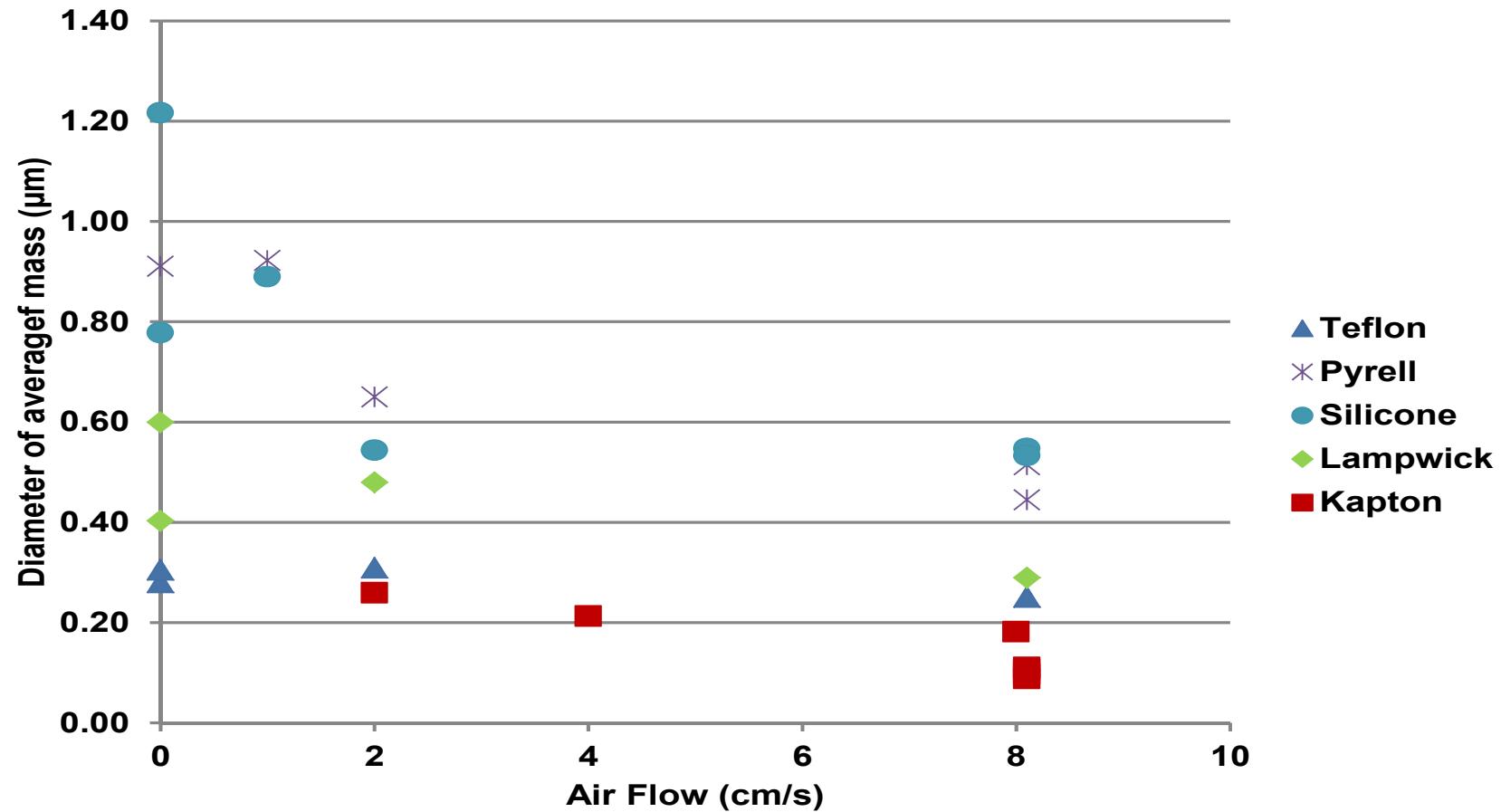
10 Micron versus 1 micron Mass ratios



10 micron versus 1 micron impactor ratios for different flow rates and sample temperatures



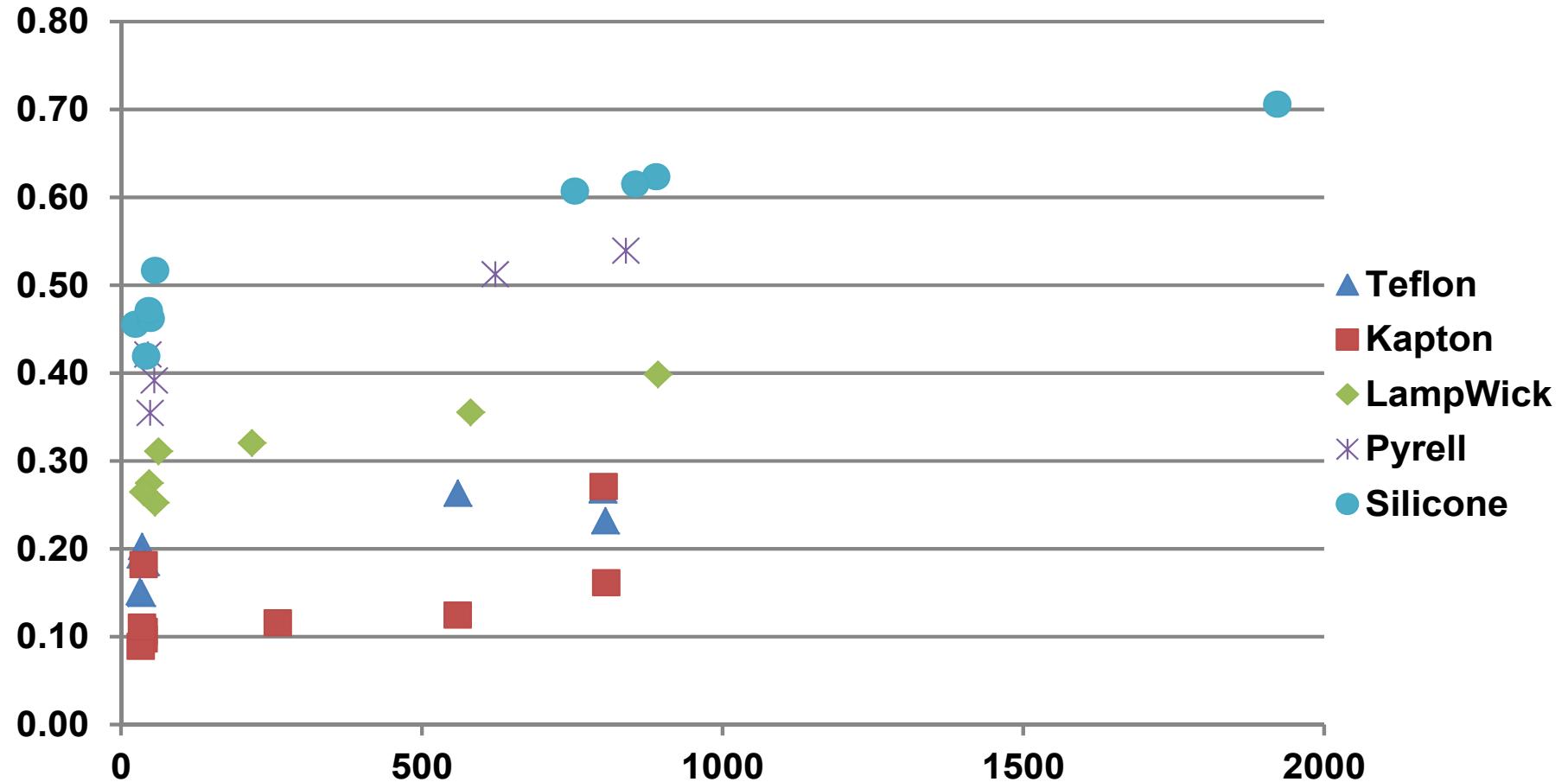
Effect of Air Flow on Diameter of Average Mass



Constant temperature for each material with no aging.



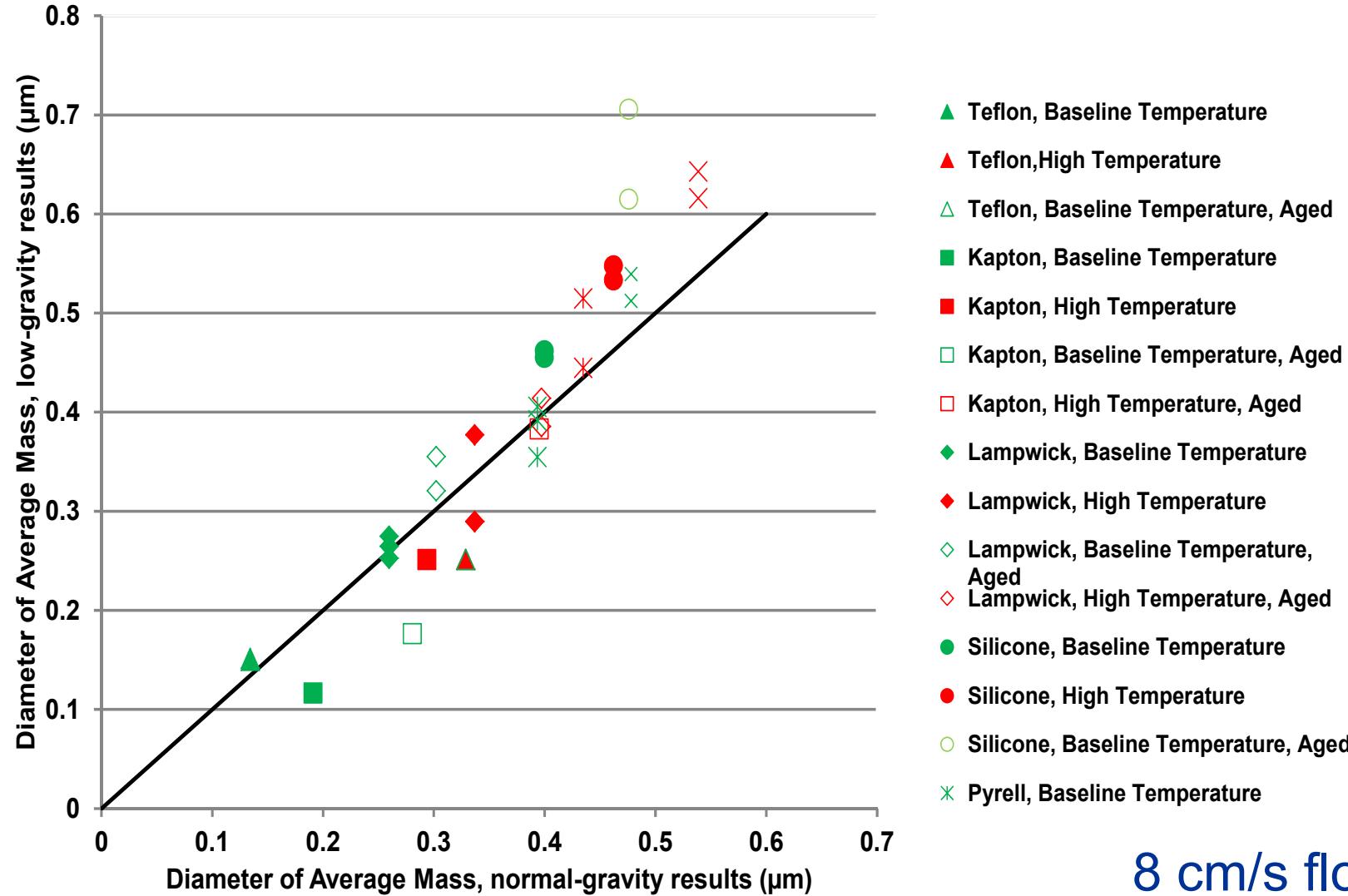
Effect of Aging on Diameter of Average Mass



8 cm/s airflow

Constant temperature for each sample.

Effect of Gravity on Diameter of Average Mass



8 cm/s flow



Conclusions

- Particle sizes ranged from 100 to 600 nm
- Consistent with a log-normal distribution.
- Particle sizes increase substantially with aging.
- Particle dimensions increase substantially as air flow was decreased.
- TEM showed a significant range of distinct particle morphologies
- For lampwick and silicone approximately 40% of the aerosol mass had aerodynamic diameters greater than 1 μm
- Ground based testing at 8 cm/s showed particle dimensions very close to the flight results.

Spacecraft fire conditions include an even wider array of materials and conditions.

Spacecraft background aerosols can be quite large

Detection methods that can measure more than one moment of the size distribution may show more successful detection and false alarm rejection than single moment detectors.